


This document set is applicable to the following part number configurations:

 <p><b>Fuel and manifold pressure systems with optional voltmeter</b></p>	<table border="1"> <thead> <tr> <th rowspan="2">Part Number</th> <th colspan="2">Kit Configuration</th> </tr> <tr> <th>Instrument</th> <th>Sensors</th> </tr> </thead> <tbody> <tr> <td>FM200</td> <td>FM200</td> <td>-</td> </tr> <tr> <td>FM200K</td> <td>FM200</td> <td>1 X A200-PSF 1 X A200-PMP</td> </tr> <tr> <td>FM200V</td> <td>FM200V</td> <td>-</td> </tr> <tr> <td>FM200VK</td> <td>FM200V</td> <td>1 X A200-PSF 1 X A200-PMP</td> </tr> </tbody> </table>	Part Number	Kit Configuration		Instrument	Sensors	FM200	FM200	-	FM200K	FM200	1 X A200-PSF 1 X A200-PMP	FM200V	FM200V	-	FM200VK	FM200V	1 X A200-PSF 1 X A200-PMP
Part Number	Kit Configuration																	
	Instrument	Sensors																
FM200	FM200	-																
FM200K	FM200	1 X A200-PSF 1 X A200-PMP																
FM200V	FM200V	-																
FM200VK	FM200V	1 X A200-PSF 1 X A200-PMP																

# FM200XX Fuel & Manifold Pressure

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## THIS IS A TRANSPORT CANADA APPROVED MANUAL

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The following checklist outlines the required articles for all FM200XX series part numbers.

	<b>Documentation</b>	<b>Document Number</b>
<input type="checkbox"/>	Introduction (this document)	S200-FM200XX-001
<input type="checkbox"/>	Warranty Statement	S200-AWS
<input type="checkbox"/>	Installation Instructions	S200-FM200XX-002
<input type="checkbox"/>	Intensity Control Installation Options	S200-DDC-INST
<input type="checkbox"/>	Wiring and Installation Schematic	S200-FM200XX-003
<input type="checkbox"/>	Human Interface Definition	S200-HID
<input type="checkbox"/>	Instrument Setup	S200-FM200XX-004
<input type="checkbox"/>	Operations Guide	S200-FM200XX-005
<input type="checkbox"/>	Product Specifications	S200-FM200XX-006
<input type="checkbox"/>	Installation Compliance	S200-FM200XX-007
<input type="checkbox"/>	Instructions for Continued Airworthiness	S200-ICA
<input type="checkbox"/>	Flight Manual Supplement	S200-FMS
<input type="checkbox"/>	Authorized Release Certificate	TCCA 24-0078
<input type="checkbox"/>	Permission to Use STC	S200-PTU
<input type="checkbox"/>	Canadian STC and Eligibility List	STC & S200-CEL
<input type="checkbox"/>	FAA STC and Approved Model List	STC & S200-AML

### Components

- FM200X Instrument
- A200-PSF Fuel Pressure Sensor
- A200-PMP Manifold Pressure Sensor
- 4 X Mounting Screws

# FM200XX Fuel & Manifold Pressure

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## The Fastest Way To Install And Start Using This Product

**NOTE:** This product is certified as a primary replacement, new or secondary instrument only when installation is performed in accordance with the documented instructions and procedures.

1. Review the Warranty Statement before performing any tasks. Do not remove the instrument from the sealed package until you agree with the stated terms.
2. If this instrument is to be installed in a certified aircraft, verify that the aircraft model is listed on the Eligibility List (Canadian registered aircraft) or the Approved Model List (USA and all other countries).

**If it is NOT listed, contact your local authority and obtain the necessary requirements or approvals before proceeding further.**

For all other vehicles, including experimental aircraft continue with Step 3.

3. Follow the Installation Instructions exactly as noted, referring to the Wiring Diagram as and when required. Note: The installation process has been optimized based on the sequence of components to be installed. Installing the product in any other way will, in all likelihood, take longer and be more complex.

Ensure that **one** of the Intensity Control options has been installed in accordance with document # S200-DDC-INST.

4. Perform the Instrument Setup using the required data as listed in the Instrument Setup procedure.
5. Validate that all functions perform as outlined in the Operations Guide.
6. Complete all regulatory documentation, if required.



## **NO NONSENSE WARRANTY**

Our warranty policy is simple .... It is even written in plain English!

**Please read it BEFORE DOING ANYTHING WITH YOUR NEW INSTRUMENT!**

**If you require technical support when installing your instrument please call our Technical Support department directly on 416-628-0725**

### **We will:**

- Repair or replace (at our discretion) any instrument which becomes defective within a period of 12 (twelve) months of manufacture date. You will pay for the shipping costs to return the instrument to us and we will pay for the shipping costs to return the instrument to you;
- Replace all instruments that fail out of warranty for a flat rate of 50% of the cost of a new instrument, at the time of the failure.

### **We are not:**

- Liable for any costs associated with the installation or removal of any of our instruments, irrespective of the cause;
- Liable for any misuse or non-use of the instrument in whatever form.

### **We will not:**

- Repair or replace your instrument free of charge, under warranty, if it has not been installed by an appropriately licensed person.

**If you do not agree with ANY of the above statements, return your new instrument to us immediately for a FULL refund LESS shipping costs.**

**ALL RETURNS REQUIRE RETURN MATERIAL AUTHORIZATIONS (RMA). WE DO NOT ACCEPT RETURNS WITHOUT RMA NUMBERS. CALL 416-628-0725 FOR AUTHORIZATION.**

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#### **Aerospace Logic Inc.**

180 James Street South, Suite 205, Hamilton, ON, L8P 4V1, CANADA  
Tel. 416-628-0725 | [www.aerospacelogic.com](http://www.aerospacelogic.com) | Fax. 416-352-5854

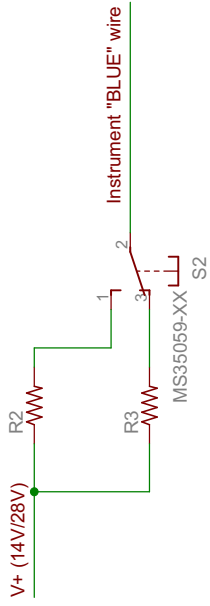
Document # S200-AWS  
Rev 1.4 12/02/2009  
Page 1 of 1

# FM200XX Installation Instructions

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1. Sensor Installation:
  - a. Install manifold pressure sensor P/N A200-PSM, and
  - b. Install fuel pressure sensor P/N A200-PSF,
  - c. in accordance with the manufacturer supplied instructions for these sensors.
2. Connect the wiring supplied with the DB25 interface connector in accordance with the FM200XX Wiring and Installation Schematic. **DO NOT ATTACH THE CONNECTOR TO THE INSTRUMENT UNTIL ALL WIRING HAS BEEN CONNECTED.**
3. Select the instrument location in the aircraft panel.
4. Ensure that the aircraft master switch and/or power to the instrument is **OFF**.
5. Attach the DB25 connector to the instrument. Ensure that the plug is fully inserted into the instrument. Finger tighten the two plug locking screws, then tighten 1/8 turn using a screwdriver. **DO NOT OVERTIGHTEN.**
6. If the instrument is to be installed to the right of the pilot's field of view, the DB25 connector on the rear of the instrument **MUST be closest to the bottom** of the instrument panel. Proceed with step 8.
7. If the instrument is to be installed in front of or to the left of the pilot's field of view the DB25 connector on the rear of the instrument **MUST be closest to the top** of the instrument panel.
8. Install the instrument in the panel, attaching with four 6-32 X 3/8" screws.
9. Initial Power Up Sequence:
  - a. Adjust the instrument intensity control for maximum intensity.
  - b. Apply power to the instrument.
  - c. The instrument display will light up.
  - d. The display may be inverted at this point. Ignore the orientation at this time.
  - e. Turn the instrument **OFF**.
10. If the instrument display does not light up:
  - a. Ensure intensity control is set for maximum.
  - b. Turn off power.
  - c. Disconnect the DB25 connector from the instrument.
  - d. Check intensity control.
  - e. Check the wiring installation and breaker or fuse.
11. Proceed to "Instrument Setup".

Fixed Intensity Installation



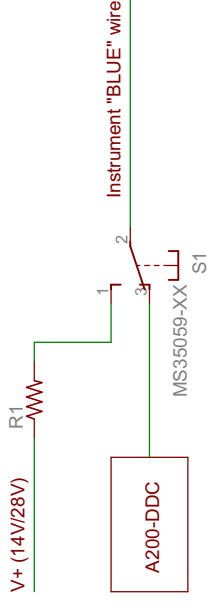
1-2 : Day/Night Operation  
 3-2 : Day Operation

R2 & R3 Part Number S200-R14-xx - 14V

# Instruments	R2	R3
1	A1	B1
2	A2	B2
3	A3	B3
4	A4	B4
5	A5	B5
6	A6	B6
7	A7	B7
8	A8	B8
9	A9	B9
10	AA	BA

R2 & R3 Part Number S200-R28-xx - 28V

Variable Intensity Installation



1-2 : Fixed Intensity  
 3-2 : Variable Intensity

R1 Part Number S200-R14-xx - 14V

# Instruments	R1
1	C1
2	C2
3	C3
4	C4
5	C5
6	C6
7	C7
8	C8
9	C9
10	CA

R1 Part Number S200-R28-xx - 28V

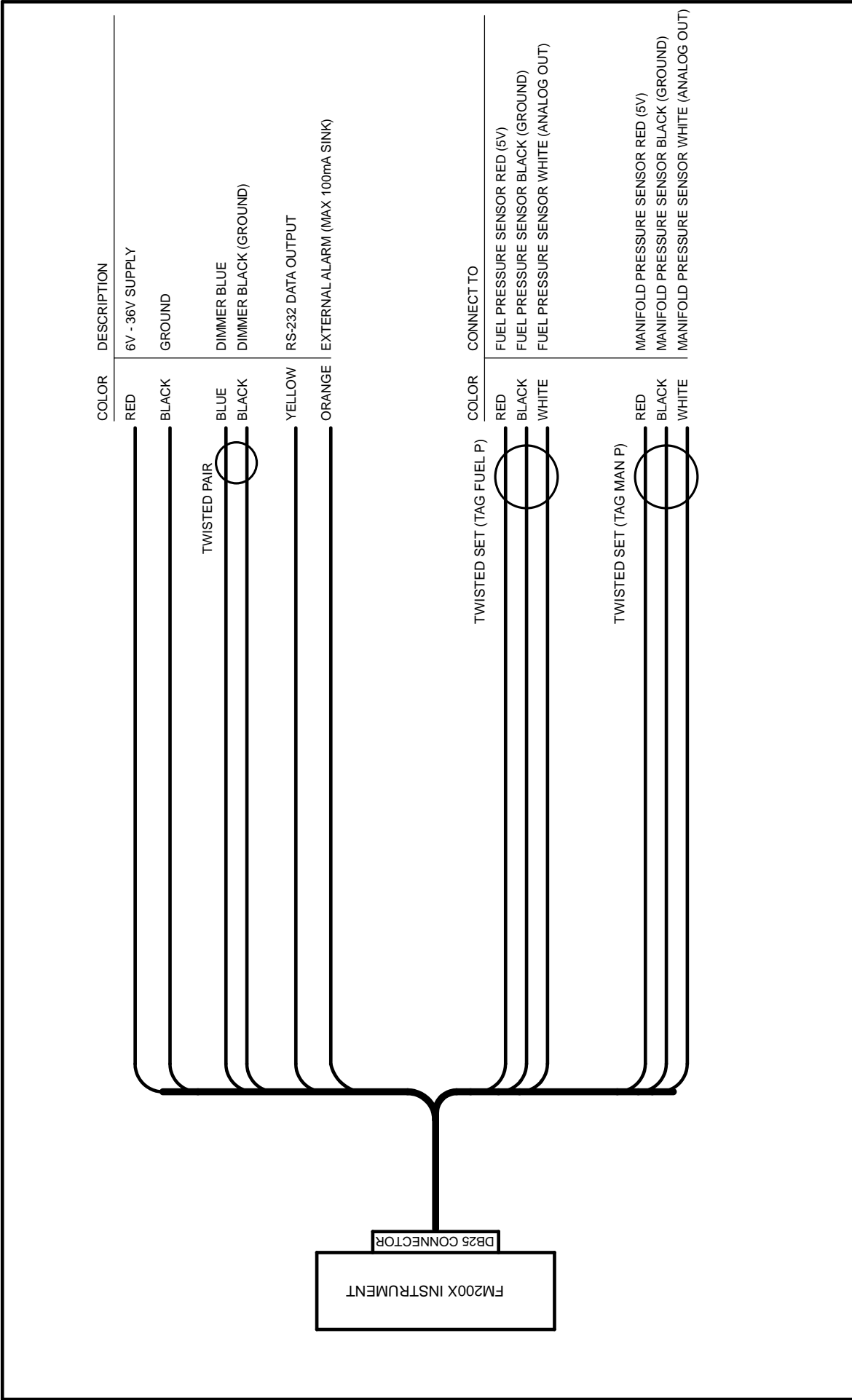
TITLE: Intensity Control - Installation Options - Rev 1.1  
 (c) 2009 Aerospace Logic Inc.

Document Number:  
 S200-DDC-INST

REV:  
 A

Date: 18/11/2009 4:24:00 PM

Sheet: 1/1



COLOR DESCRIPTION

RED 6V - 36V SUPPLY

BLACK GROUND

BLUE DIMMER BLUE  
 BLACK DIMMER BLACK (GROUND)

YELLOW RS-232 DATA OUTPUT

ORANGE EXTERNAL ALARM (MAX. 100mA SINK)

COLOR CONNECT TO

RED FUEL PRESSURE SENSOR RED (5V)

BLACK FUEL PRESSURE SENSOR BLACK (GROUND)

WHITE FUEL PRESSURE SENSOR WHITE (ANALOG OUT)

RED MANIFOLD PRESSURE SENSOR RED (5V)

BLACK MANIFOLD PRESSURE SENSOR BLACK (GROUND)

WHITE MANIFOLD PRESSURE SENSOR WHITE (ANALOG OUT)

INSTALLATION NOTES:

1. Allow for current drain of 0.1A (100mA)
2. Install using breaker or panel fuse
3. Maximum 7 instruments per 1A of breaker capacity
4. For multiple instrument installations install breakers to comply with redundancy requirements of AC 23.1309-1D Appendix 1
5. Connect all BLACK (GROUND) wires to the same ground point

ACCESSORY NOTES:

1. For dimmer use LCD Lighting Controller P/N A200-DDC
2. For manifold pressure sensor use P/N A200-PMP
3. For fuel pressure sensor use P/N A200-PSF

# FM200XX Wiring and Installation Schematic

# 200 Series Human Interface Definition

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All 200 Series instruments have been designed around a standard Human Interface protocol. This allows identical operation of the two panel switches between all instruments while still allowing multiple functions. This document is a generic outline of the functionality of the switches, based on the various functional states of the instrument.

## Top Button

The top button is the action button. It invokes different screen displays, allows selection of values and increments or decrements values.

Examples of these actions are:

1. Moving between display screens
2. Selection of an action from a menu
3. Changing a value from within the setup menus
4. Entering a value during flight (clock instrument only)

## Bottom Button

The bottom button is the navigation button. It will allow the cursor on the screen to move between allowable selections.

Examples of these actions are:

1. Selection of different cylinders (EGT and CHT) in the Single Cylinder mode
2. Selection of different tanks (Fuel Level)
3. Selection of limits and functional items from within the setup menus
4. Time field selection (clock instrument only)

There is one exception to the functionality of the bottom button. This is from the Primary Display (the screen that first comes on from power up). Pressing the button will display the instrument core status and support data.

## General

1. Each specific instrument contains detailed button options within the Operations Guide.
2. During setup, instructions are provided on screen and use of the buttons is as noted in this document.



# FM200XX Instrument Setup

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1. Before proceeding with the instrument setup you will require the following information, as it pertains to the aircraft that the instrument is being installed on:
  - a. Minimum operating fuel pressure
  - b. Maximum operating fuel pressure
  - c. Fuel pressure redline
  
  - d. Minimum operating manifold pressure
  - e. Maximum operating manifold pressure
  - f. Manifold pressure redline
  
2. This information must be obtained from a current, traceable source for the aircraft that the instrument is being installed on. Some possible sources are:
  - a. Pilot Operating Handbook
  - b. Maintenance Manual
  - c. Aircraft Type Certificate Data Sheet
  - d. Engine Type Certificate Data Sheet
  - e. Manufacturer
  
3. Then, enter the correct parameters into the instrument as follows:
  - a. Turn off the power to the instrument
  - b. Press and hold both top and bottom buttons and turn on the power to the instrument.
  - c. When instructed, release both buttons.
  - d. In the setup menus, use the bottom button to navigate and the top button to select an option.
  - e. Follow the onscreen instructions. **NOTE: Depending on the installation orientation of the instrument the display may initially be upside down. The first prompt in the setup is to press the button at the top of the instrument. This will orientate it correctly.** Continue with setup.

# FM200XX Instrument Setup

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4. Once all steps of the setup have been completed, the instrument will provide the option to edit any data or to save and exit.
5. If the data entered is correct, choose the "OK" option to save the data and exit setup, otherwise choose "EDIT".
6. Once the instrument has exited the setup it will restart. It is now ready for use.

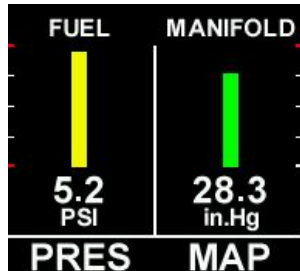
# FM200XX Operations Guide

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This manual outlines the operating procedures for all FM200XX series products

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## 1. Power Up / Default Screen



Displays the fuel and manifold pressure information. Colors are determined by the range values as set during installation.

Press the top button once to move to the next display screen (#2 if the part number contains a "V" suffix **OR** #3 if not).

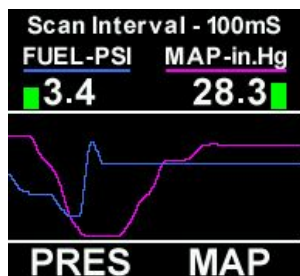
## 2. Voltmeter Display



Displays the battery voltage on the top half of the display and the fuel and manifold pressure information on the bottom half.

Press the top button to move to the next display.

## 3. Trend Graph



This screen provides a graphical display of the fuel and manifold pressure.

Both data sets are displayed with a miniature version of the bars as on the default screen and the actual value.

The "Scan Interval" can be changed by pressing the bottom button.

To freeze the display, press the top button **ONCE**.

To continue displaying information in this mode, press the bottom button **ONCE**, **OR**, to return to the default screen, press the top button once.

# FM200XX Operations Guide

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## 4. Other Options

To display the instrument serial number and core information press and hold the bottom button when in the default screen.

### WARNING

**The FM200 instrument may only be installed as a primary instrument when an exhaust gas temperature, cylinder head temperature or fuel flow instrument is installed (any manufacturer) in the aircraft.**

**This is a requirement in meeting the failure mode criteria of AC 23.1309-1D Appendix 1**

### WARNING

*All engine operations are to be performed in accordance with the specific instructions pertaining to your aircraft, including those provided by the engine manufacturer. The FM200XX series instruments are only parameter reporting devices which can assist in engine management when used in conjunction with these instructions.*

*All FM200XX products are certified as primary replacements, within the limits as described above, or for secondary use as desired.*

# FM200XX Product Specifications

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## Certification / Compliance

TSO	C47a and C45b
Environmental	DO-160F
	A1CAASXXXXXXXXZBXXBBBCSBA1C11XXAX
Software	DO-178B

## DC Power Source

Input voltage	6 to 36 VDC
Power consumption	100mA
Load dump tolerance	+60V
Direct spike tolerance	+/- 60V
Cable spike tolerance	> +/- 1KV

## Operating Temperature

Constant operating	-15°C to +55°C
Short term operating (1hr)	-40°C to +70°C
Storage	-55°C to +85°C

## Signal Input

Pressure	0.5V to 3.0V
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## Display

Fuel pressure	0.0 PSI to 90.0 PSI
Manifold pressure	0.0 to 50.0 inHg
Voltmeter	6.0V to 36V

## Accuracy

Fuel pressure AS408C	+/- 0.25 PSI
Manifold pressure AS8042	+/- 0.1 inHg
Voltmeter	+/- 0.1V

## Display

Sunlight readable LCD  
176 X 192 display pixels, 65,535 colors

## Annunciator Output

100mA sink with reverse EMF protection

## RS-232 Data Output

1 second per full sequence at 57.6K bps (8/N/2/N)

## Interface Connector

DB-25

## Software Functionality

Primary	Fuel pressure, manifold pressure
Voltmeter	Voltmeter, fuel and manifold pressure
Trend Graph	Graphic display of fuel and manifold pressure

## Dimensions

2.45" X 2.45" X 0.9"

## Weight (excluding harness)

4.05 oz

## Finish

Black anodized 6061 aluminum

# FM200XX Installation Compliance

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## 1. Installation Procedures and Limitations

Should your specific aircraft not be listed on our STC Approved Model List (AML), contact your local FSDO or flight authority for installation requirements **BEFORE commencing any installation tasks.**

Installation must be in accordance with the following FM200XX published documents:

- a. Installation Instructions,
- b. Wiring and Installation Schematic, and
- c. Instrument Setup.

## 2. TSO COMPLIANCE NOTICE

The conditions and tests required for TSO approval of this instrument are minimum performance standards. Those installing this instrument on or in a specific type or class of aircraft, must determine that the aircraft installation conditions are within the TSO standards. TSO articles must have separate approval for installation in an aircraft. The instrument may be installed only according to 14 CFR part 43 or the applicable airworthiness requirement.