





This document set is applicable to the following part number configurations:

 <p>Four cylinder CHT systems and probe configurations</p>	<table border="1"> <thead> <tr> <th rowspan="2">Part Number</th> <th colspan="2">Kit Configuration</th> </tr> <tr> <th>Instrument</th> <th>CHT Probe</th> </tr> </thead> <tbody> <tr> <td>CT204</td> <td>CT204</td> <td>-</td> </tr> <tr> <td>CT204B</td> <td>CT204</td> <td>4 X A200-TCB</td> </tr> <tr> <td>CT204R2</td> <td>CT204</td> <td>4 X A200-TCR-12</td> </tr> <tr> <td>CT204R4</td> <td>CT204</td> <td>4 X A200-TCR-14</td> </tr> <tr> <td>CT204R8</td> <td>CT204</td> <td>4 X A200-TCR-18</td> </tr> </tbody> </table>	Part Number	Kit Configuration		Instrument	CHT Probe	CT204	CT204	-	CT204B	CT204	4 X A200-TCB	CT204R2	CT204	4 X A200-TCR-12	CT204R4	CT204	4 X A200-TCR-14	CT204R8	CT204	4 X A200-TCR-18
Part Number	Kit Configuration																				
	Instrument	CHT Probe																			
CT204	CT204	-																			
CT204B	CT204	4 X A200-TCB																			
CT204R2	CT204	4 X A200-TCR-12																			
CT204R4	CT204	4 X A200-TCR-14																			
CT204R8	CT204	4 X A200-TCR-18																			
 <p>Six cylinder CHT systems and probe configurations</p>	<table border="1"> <thead> <tr> <th rowspan="2">Part Number</th> <th colspan="2">Kit Configuration</th> </tr> <tr> <th>Instrument</th> <th>CHT Probe</th> </tr> </thead> <tbody> <tr> <td>CT206</td> <td>CT206</td> <td>-</td> </tr> <tr> <td>CT206B</td> <td>CT206</td> <td>6 X A200-TCB</td> </tr> <tr> <td>CT206R2</td> <td>CT206</td> <td>6 X A200-TCR-12</td> </tr> <tr> <td>CT206R4</td> <td>CT206</td> <td>6 X A200-TCR-14</td> </tr> <tr> <td>CT206R8</td> <td>CT206</td> <td>6 X A200-TCR-18</td> </tr> </tbody> </table>	Part Number	Kit Configuration		Instrument	CHT Probe	CT206	CT206	-	CT206B	CT206	6 X A200-TCB	CT206R2	CT206	6 X A200-TCR-12	CT206R4	CT206	6 X A200-TCR-14	CT206R8	CT206	6 X A200-TCR-18
Part Number	Kit Configuration																				
	Instrument	CHT Probe																			
CT206	CT206	-																			
CT206B	CT206	6 X A200-TCB																			
CT206R2	CT206	6 X A200-TCR-12																			
CT206R4	CT206	6 X A200-TCR-14																			
CT206R8	CT206	6 X A200-TCR-18																			
 <p>Eight cylinder CHT systems and probe configurations</p>	<table border="1"> <thead> <tr> <th rowspan="2">Part Number</th> <th colspan="2">Kit Configuration</th> </tr> <tr> <th>Instrument</th> <th>CHT Probe</th> </tr> </thead> <tbody> <tr> <td>CT208</td> <td>CT208</td> <td>-</td> </tr> <tr> <td>CT208B</td> <td>CT208</td> <td>8 X A200-TCB</td> </tr> <tr> <td>CT208R2</td> <td>CT208</td> <td>8 X A200-TCR-12</td> </tr> <tr> <td>CT208R4</td> <td>CT208</td> <td>8 X A200-TCR-14</td> </tr> <tr> <td>CT208R8</td> <td>CT208</td> <td>8 X A200-TCR-18</td> </tr> </tbody> </table>	Part Number	Kit Configuration		Instrument	CHT Probe	CT208	CT208	-	CT208B	CT208	8 X A200-TCB	CT208R2	CT208	8 X A200-TCR-12	CT208R4	CT208	8 X A200-TCR-14	CT208R8	CT208	8 X A200-TCR-18
Part Number	Kit Configuration																				
	Instrument	CHT Probe																			
CT208	CT208	-																			
CT208B	CT208	8 X A200-TCB																			
CT208R2	CT208	8 X A200-TCR-12																			
CT208R4	CT208	8 X A200-TCR-14																			
CT208R8	CT208	8 X A200-TCR-18																			
 <p>Nine cylinder CHT systems and probe configurations</p>	<table border="1"> <thead> <tr> <th rowspan="2">Part Number</th> <th colspan="2">Kit Configuration</th> </tr> <tr> <th>Instrument</th> <th>CHT Probe</th> </tr> </thead> <tbody> <tr> <td>CT209</td> <td>CT209</td> <td>-</td> </tr> <tr> <td>CT209B</td> <td>CT209</td> <td>9 X A200-TCB</td> </tr> <tr> <td>CT209R2</td> <td>CT209</td> <td>9 X A200-TCR-12</td> </tr> <tr> <td>CT209R4</td> <td>CT209</td> <td>9 X A200-TCR-14</td> </tr> <tr> <td>CT209R8</td> <td>CT209</td> <td>9 X A200-TCR-18</td> </tr> </tbody> </table>	Part Number	Kit Configuration		Instrument	CHT Probe	CT209	CT209	-	CT209B	CT209	9 X A200-TCB	CT209R2	CT209	9 X A200-TCR-12	CT209R4	CT209	9 X A200-TCR-14	CT209R8	CT209	9 X A200-TCR-18
Part Number	Kit Configuration																				
	Instrument	CHT Probe																			
CT209	CT209	-																			
CT209B	CT209	9 X A200-TCB																			
CT209R2	CT209	9 X A200-TCR-12																			
CT209R4	CT209	9 X A200-TCR-14																			
CT209R8	CT209	9 X A200-TCR-18																			

# CT20XXX Cylinder Head Temperature

---

**THIS IS A TRANSPORT CANADA APPROVED MANUAL**

---

The following checklist outlines the required articles for all CT20XXX series part numbers.

	<b>Documentation</b>	<b>Document Number</b>
<input type="checkbox"/>	Introduction (this document)	S200-CT20XXX-001
<input type="checkbox"/>	Warranty Statement	S200-AWS
<input type="checkbox"/>	Installation Instructions	S200-CT20XXX-002
<input type="checkbox"/>	Intensity Control Installation Options	S200-DDC-INST
<input type="checkbox"/>	Wiring and Installation Schematic	S200-CT20XXX-003
<input type="checkbox"/>	Human Interface Definition	S200-HID
<input type="checkbox"/>	Instrument Setup	S200-CT20XXX-004
<input type="checkbox"/>	Operations Guide	S200-CT20XXX-005
<input type="checkbox"/>	Product Specifications	S200-CT20XXX-006
<input type="checkbox"/>	Installation Compliance	S200-CT20XXX-007
<input type="checkbox"/>	Instructions for Continued Airworthiness	S200-ICA
<input type="checkbox"/>	Flight Manual Supplement	S200-FMS
<input type="checkbox"/>	Authorized Release Certificate	TCCA 24-0078
<input type="checkbox"/>	Permission to Use STC	S200-PTU
<input type="checkbox"/>	Canadian STC and Eligibility List	STC & S200-CEL
<input type="checkbox"/>	FAA STC and Approved Model List	STC & S200-AML

## **Components**

- CT20X Instrument
- Thermocouple kit including installation hardware
- 4 X Mounting Screws

# CT20XXX Cylinder Head Temperature

---

## The Fastest Way To Install And Start Using This Product

**NOTE:** This product is certified as a primary replacement, new or secondary instrument only when installation is performed in accordance with the documented instructions and procedures.

1. Review the Warranty Statement before performing any tasks. Do not remove the instrument from the sealed package until you agree with the stated terms.
2. If this instrument is to be installed in a certified aircraft, verify that the aircraft model is listed on the Eligibility List (Canadian registered aircraft) or the Approved Model List (USA and all other countries).

**If it is NOT listed, contact your local authority and obtain the necessary requirements or approvals before proceeding further.**

For all other vehicles, including experimental aircraft continue with Step 3.

3. Follow the Installation Instructions exactly as noted, referring to the Wiring Diagram as and when required. Note: The installation process has been optimized based on the sequence of components to be installed. Installing the product in any other way will, in all likelihood, take longer and be more complex.

Ensure that **one** of the Intensity Control options has been installed in accordance with document # S200-DDC-INST.

4. Perform the Instrument Setup using the required data as listed in the Instrument Setup procedure.
5. Validate that all functions perform as outlined in the Operations Guide.
6. Complete all regulatory documentation, if required.

---

### Aerospace Logic Inc.

180 James Street South, Suite 205, Hamilton, ON, L8P 4V1, CANADA  
Tel. 416-628-0725 | www.aerospacelogic.com | Fax. 416-352-5854

Document # S200-CT20XXX-001  
Rev 1.1 12/03/2009  
Page 2 of 2



## **NO NONSENSE WARRANTY**

Our warranty policy is simple .... It is even written in plain English!

**Please read it BEFORE DOING ANYTHING WITH YOUR NEW INSTRUMENT!**

**If you require technical support when installing your instrument please call our Technical Support department directly on 416-628-0725**

### **We will:**

- Repair or replace (at our discretion) any instrument which becomes defective within a period of 12 (twelve) months of manufacture date. You will pay for the shipping costs to return the instrument to us and we will pay for the shipping costs to return the instrument to you;
- Replace all instruments that fail out of warranty for a flat rate of 50% of the cost of a new instrument, at the time of the failure.

### **We are not:**

- Liable for any costs associated with the installation or removal of any of our instruments, irrespective of the cause;
- Liable for any misuse or non-use of the instrument in whatever form.

### **We will not:**

- Repair or replace your instrument free of charge, under warranty, if it has not been installed by an appropriately licensed person.

**If you do not agree with ANY of the above statements, return your new instrument to us immediately for a FULL refund LESS shipping costs.**

**ALL RETURNS REQUIRE RETURN MATERIAL AUTHORIZATIONS (RMA). WE DO NOT ACCEPT RETURNS WITHOUT RMA NUMBERS. CALL 416-628-0725 FOR AUTHORIZATION.**

---

#### **Aerospace Logic Inc.**

180 James Street South, Suite 205, Hamilton, ON, L8P 4V1, CANADA  
Tel. 416-628-0725 | [www.aerospacelogic.com](http://www.aerospacelogic.com) | Fax. 416-352-5854

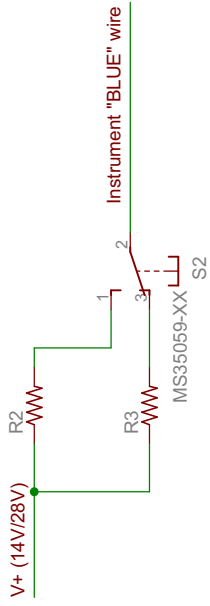
Document # S200-AWS  
Rev 1.4 12/02/2009  
Page 1 of 1

# CT20XXX Installation Instructions

---

1. Install thermocouple probes in accordance with the applicable manufacturer instructions. **Use only Type “J” ungrounded probes.**
2. Connect the wiring supplied with the DB25 interface connector in accordance with the CT20XXX Wiring and Installation Schematic. **DO NOT ATTACH THE CONNECTOR TO THE INSTRUMENT UNTIL ALL WIRING HAS BEEN CONNECTED.**
3. Select the instrument location in the aircraft panel.
4. Ensure that the aircraft master switch and/or power to the instrument is **OFF**.
5. Attach the DB25 connector to the instrument. Ensure that the plug is fully inserted into the instrument. Finger tighten the two plug locking screws, then tighten 1/8 turn using a screwdriver. **DO NOT OVERTIGHTEN.**
6. If the instrument is to be installed to the right of the pilot’s field of view, the DB25 connector on the rear of the instrument **MUST be closest to the bottom** of the instrument panel. Proceed with step 8.
7. If the instrument is to be installed in front of or to the left of the pilot’s field of view the DB25 connector on the rear of the instrument **MUST be closest to the top** of the instrument panel.
8. Install the instrument in the panel, attaching with four 6-32 X 3/8” screws.
9. Initial Power Up Sequence:
  - a. Adjust the instrument intensity control for maximum intensity.
  - b. Apply power to the instrument.
  - c. The instrument display will light up.
  - d. The display may be inverted at this point. Ignore the orientation at this time.
  - e. Turn the instrument **OFF**.
10. If the instrument display does not light up:
  - a. Ensure intensity control is set for maximum.
  - b. Turn off power.
  - c. Disconnect the DB25 connector from the instrument.
  - d. Check intensity control.
  - e. Check the wiring installation and breaker or fuse.
11. Proceed to “Instrument Setup”.

Fixed Intensity Installation



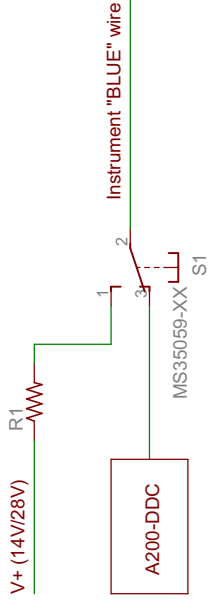
1-2 : Day/Night Operation  
3-2 : Day Operation

R2 & R3 Part Number S200-R14-xx - 14V

# Instruments	R2	R3
1	A1	B1
2	A2	B2
3	A3	B3
4	A4	B4
5	A5	B5
6	A6	B6
7	A7	B7
8	A8	B8
9	A9	B9
10	AA	BA

R2 & R3 Part Number S200-R28-xx - 28V

Variable Intensity Installation



1-2 : Fixed Intensity  
3-2 : Variable Intensity

R1 Part Number S200-R14-xx - 14V

# Instruments	R1
1	C1
2	C2
3	C3
4	C4
5	C5
6	C6
7	C7
8	C8
9	C9
10	CA

R1 Part Number S200-R28-xx - 28V

**TITLE:** Intensity Control - Installation Options - Rev 1.1  
(c) 2009 Aerospace Logic Inc.

**Document Number:**  
S200-DDC-INST

**REV:**  
A

**Date:** 18/11/2009 4:24:00 PM

**Sheet:** 1/1



COLOR	DESCRIPTION
RED	6V - 36V SUPPLY
BLACK	GROUND
BLUE	DIMMER BLUE
BLACK	DIMMER BLACK (GROUND)
YELLOW	RS-232 DATA OUTPUT
ORANGE	EXTERNAL ALARM (MAX. 100mA SINK)

TAG #	THERMOCOUPLE # BY INSTRUMENT MODEL
1	ALL #1
2	ALL #2
3	ALL #3
4	ALL #4
5	CT206/8/9 (#5)
6	CT206/8/9 (#6)
7	CT208/9 (#7)
8	CT208/9 (#8)
9	CT209 (#9)

**INSTALLATION NOTES:**

- Allow for current drain of 0.1A (100mA)
- Install using breaker or panel fuse
- Maximum 7 instruments per 1A of breaker capacity
- For multiple instrument installations install breakers to comply with redundancy requirements of AC 23.1309-1D Appendix 1
- Connect all BLACK (GROUND) wires to the same ground point

**ACCESSORY NOTES:**

- For dimmer use LCD Lighting Controller P/N A200-DDC
- Thermocouples are ungrounded Type "J"
- Thermocouple V+ (WHITE) to large connector
- Thermocouple V- (RED) to small connector
- Thermocouple P/N's A200-TCB / -TCR12 / -TCR14 / -TCR18

# 200 Series Human Interface Definition

---

All 200 Series instruments have been designed around a standard Human Interface protocol. This allows identical operation of the two panel switches between all instruments while still allowing multiple functions. This document is a generic outline of the functionality of the switches, based on the various functional states of the instrument.

## Top Button

The top button is the action button. It invokes different screen displays, allows selection of values and increments or decrements values.

Examples of these actions are:

1. Moving between display screens
2. Selection of an action from a menu
3. Changing a value from within the setup menus
4. Entering a value during flight (clock instrument only)

## Bottom Button

The bottom button is the navigation button. It will allow the cursor on the screen to move between allowable selections.

Examples of these actions are:

1. Selection of different cylinders (EGT and CHT) in the Single Cylinder mode
2. Selection of different tanks (Fuel Level)
3. Selection of limits and functional items from within the setup menus
4. Time field selection (clock instrument only)

There is one exception to the functionality of the bottom button. This is from the Primary Display (the screen that first comes on from power up). Pressing the button will display the instrument core status and support data.

## General

1. Each specific instrument contains detailed button options within the Operations Guide.
2. During setup, instructions are provided on screen and use of the buttons is as noted in this document.



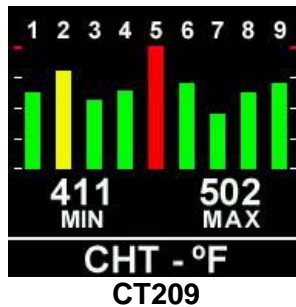
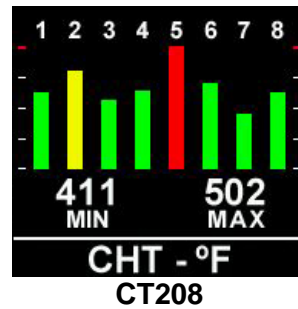
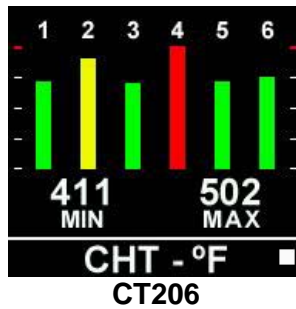
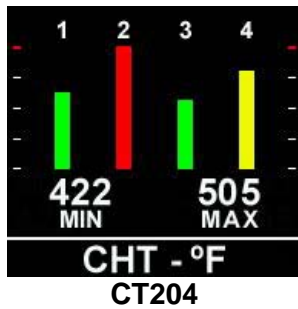
# CT20XXX Instrument Setup

---

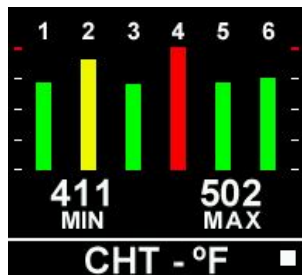
1. Before proceeding with the instrument setup you will require the following information, as it pertains to the aircraft that the instrument is being installed on:
  - a. Minimum operating CHT temperature
  - b. Maximum operating CHT temperature
  - c. CHT redline temperature
2. This information must be obtained from a current, traceable source for the aircraft that the instrument is being installed on. Some possible sources are:
  - a. Pilot Operating Handbook
  - b. Maintenance Manual
  - c. Aircraft Type Certificate Data Sheet
  - d. Engine Type Certificate Data Sheet
  - e. Manufacturer
3. Then, enter the correct parameters into the instrument as follows:
  - a. Turn off the power to the instrument
  - b. Press and hold both top and bottom buttons and turn on the power to the instrument.
  - c. When instructed, release both buttons.
  - d. In the setup menus, use the bottom button to navigate and the top button to select an option.
  - e. Follow the onscreen instructions. **NOTE: Depending on the installation orientation of the instrument the display may initially be upside down. The first prompt in the setup is to press the button at the top of the instrument. This will orientate it correctly.** Continue with setup.
4. Once all steps of the setup have been completed, the instrument will provide the option to edit any data or to save and exit.
5. If the data entered is correct, choose the “OK” option to save the data and exit setup, otherwise choose “EDIT”.
6. Once the instrument has exited the setup it will restart. It is now ready for use.

# CT20XXX Operations Guide

This manual outlines the operating procedures for all CT20XXX series products



## 1. Power Up / Default MinMax Screen

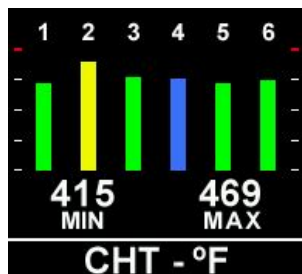


Displays the temperature of all cylinders in a graphic format. Colors are determined by the values set during installation.

Displays the minimum and maximum temperatures in the selected units.

The white square character will flash in the right corner for the first 60 seconds of operation.

During this time the instrument is gathering data for the Shock Cooling function. Once the flashing block disappears, the Shock Cooling function is available. This only takes place when the power is first applied to the instrument or the instrument resets due to a power fallout.



If the temperature of one or more cylinders changes by an amount greater than the preset shock cooling value, then the specific cylinder indication will turn blue and flash.

Allow the cylinder/s to warm (add power) to bring the engine out of the shock cooling condition.

### Aerospace Logic Inc.

180 James Street South, Suite 205, Hamilton, ON, L8P 4V1, CANADA  
Tel. 416-628-0725 | www.aerospacelogic.com | Fax. 416-352-5854

Document # S200-CT20XXX-005  
Rev 1.2 12/13/2009  
Page 1 of 3

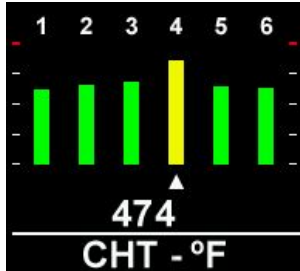
# CT20XXX Operations Guide

---

## 2. Single Cylinder

To monitor the temperature on one specific cylinder, press the top button once when in the primary MinMax display mode.

The following screen will be displayed:



Use the bottom button to scroll to the desired cylinder (default is 1).

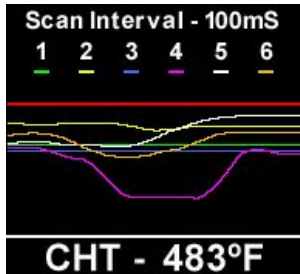
The temperature of the selected cylinder will be displayed.

When desired, press the top button to return to the MinMax screen.

## 3. Trend Graph

To monitor CHT temperatures graphically, press the top button once from within the Single Cylinder display mode.

The following screen will be displayed:



The temperature displayed is that of the hottest cylinder.

The "Scan Interval" can be changed by pressing the bottom button.

To freeze the display, press the top button **ONCE**.

To continue displaying temperatures in this mode, press the bottom button **ONCE**, OR, to return to the Menu screen, press the top button once.

# CT20XXX Operations Guide

---

## 4. Other Options

To display the instrument serial number and core information press and hold the bottom button when in the MinMax screen.

### **WARNING**

*All engine operations are to be performed in accordance with the specific instructions pertaining to your aircraft, including those provided by the engine manufacturer. The CT20XXX series instruments are only parameter reporting devices which can assist in engine management when used in conjunction with these instructions.*

*All CT20XXX products are certified as primary replacements, or for secondary use as desired.*

# CT20XXX Product Specifications

---

## Certification / Compliance

TSO	C43c
Environmental	DO-160F A1CAASXXXXXXXXZBXXBBBCSBA1C11XXAX
Software	DO-178B

## DC Power Source

Input voltage	6 to 36 VDC
Power consumption	100mA
Load dump tolerance	+60V
Direct spike tolerance	+/- 60V
Cable spike tolerance	> +/- 1KV

## Operating Temperature

Constant operating	-15°C to +55°C
Short term operating (1hr)	-40°C to +70°C
Storage	-55°C to +85°C

## Signal Input

0.0V to 20.255mV, Type-J thermocouple  
4 – 9 channels, defined by part number

## Temperature Display

100°F to 700°F

## Accuracy

SAE AS8005 Class IIa (+/- 1% FS)

## Display Units

°F or °C, user selectable

## Display

Sunlight readable LCD  
176 X 192 display pixels, 65,535 colors

## Annunciator Output

100mA sink with reverse EMF protection

## RS-232 Data Output

1 second per full sequence at 57.6K bps (8/N/2/N)

## Interface Connector

DB-25

## Software Functionality

All screens	Shock cooling sense
Primary	All cylinders with Min and Max temperatures
Single Cylinder	Display all cylinders, but monitor one
Trend Graph	Graphic display of temperatures

## Dimensions

2.45" X 2.45" X 0.9"

## Weight (excluding harness)

4.05 oz

## Finish

Black anodized 6061 aluminum

# CT20XXX Installation Compliance

---

## 1. Installation Procedures and Limitations

Should your specific aircraft not be listed on our STC Approved Model List (AML), contact your local FSDO or flight authority for installation requirements **BEFORE commencing any installation tasks.**

Installation must be in accordance with the following CT20XXX published documents:

- a. Installation Instructions,
- b. Wiring and Installation Schematic, and
- c. Instrument Setup.

## 2. TSO COMPLIANCE NOTICE

The conditions and tests required for TSO approval of this instrument are minimum performance standards. Those installing this instrument on or in a specific type or class of aircraft, must determine that the aircraft installation conditions are within the TSO standards. TSO articles must have separate approval for installation in an aircraft. The instrument may be installed only according to 14 CFR part 43 or the applicable airworthiness requirement.